

EXOMARS

- Two missions are foreseen within the ExoMars programme:
 - 2016: Orbiter plus an Entry, Descent and Landing Demonstrator Module
 - 2018: Rover
- Both missions carried out in cooperation with Roscosmos
- The ExoMars programme will demonstrate a number of essential flight and in-situ enabling technologies necessary for future exploration missions, such as an international Mars Sample Return mission
- At the same time a number of important scientific investigations will be carried out, as the search for signs of past and present life on





2016 Mission

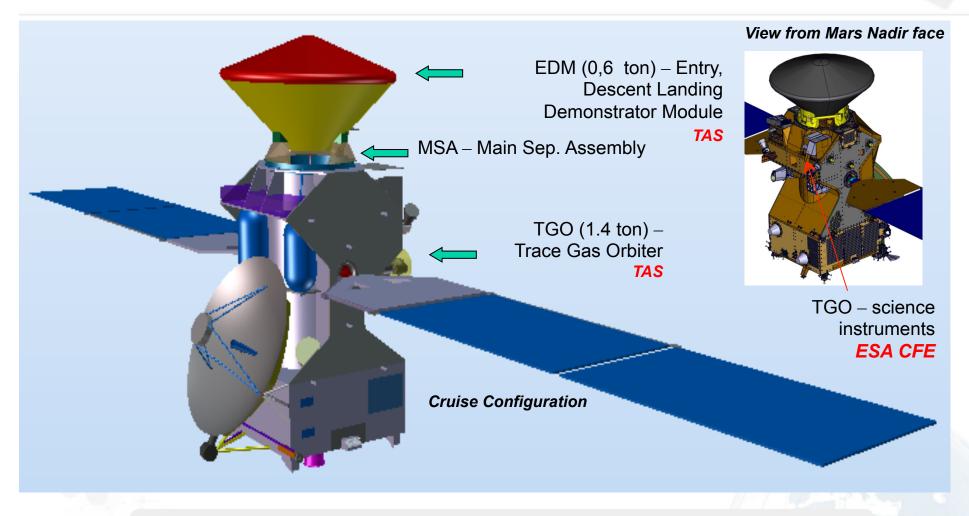
2016 Mission Objectives

- Technological Objectives
 - To demonstrate European ability to land a surface package on Mars
 - Entry, Descent and Landing Demonstrator Module (EDM)
 - DM Release: From hyperbolic trajectory, Oct 2016 Landing Accuracy: 50 km 3σ, ellipse major axis dispersion
- Scientific Objectives
 - Characterise Martian atmosphere gases
 - Trace Gas Orbiter (TGO) and its 4-Experiments package
- **Data Relay**
 - Provide communications link to ground for 2018 EXOMARS mission and other future Martian missions
 - **UHF package inside TGO**

4/16/14

20 January 2012

2016 mission - Spacecraft Composite



TAS-I Torino: Overall Prime Contractor

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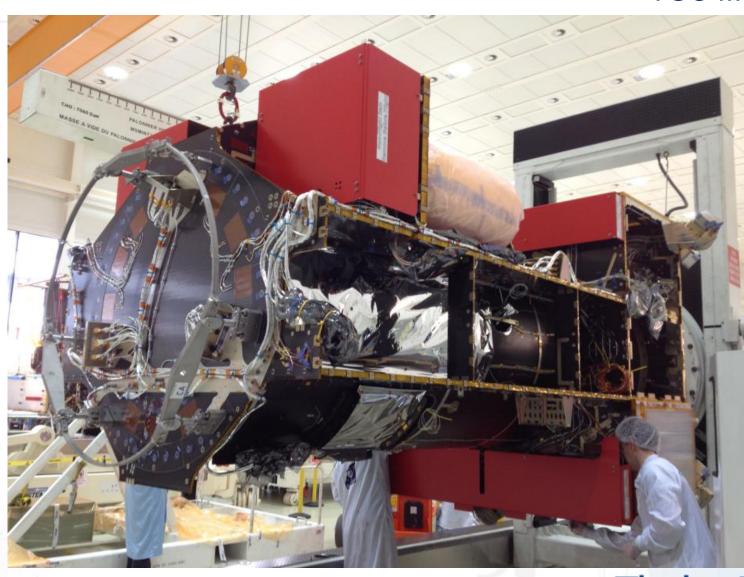
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TGO MTP

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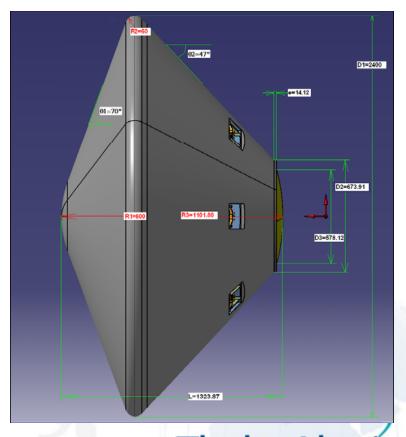
Turin December 4th, 2013

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2016 EDL Demonstrator Module (EDM)

- Blunt-shaped vehicle:
 - 70° sphere-cone Front Shield
 - → 47° conical Back Shield
 - Break-Out Patch
- EDM Outer diameter: 2.4 m
- EDM Height: 1.3 m
- EDM Mass: 600 kg
- EDM CoG distance from nose <27% diameter</p>





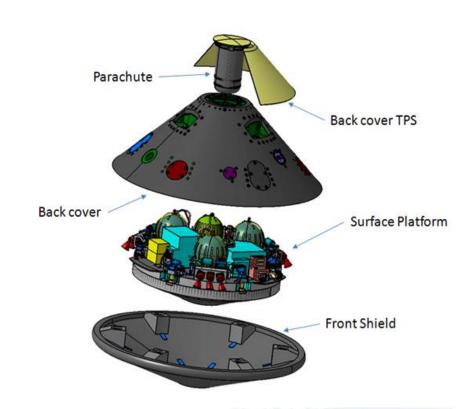




2016 EDL Demonstrator Module (EDM)

Demonstrate European entry, descent and landing technologies for future missions to Mars

- Parachute deployment (Mach 1.95)
- ➤ Front Shield jettison → RDA operational
- Back Cover separation (1400 m, 80 m/s) → RCS activation 1 s later
- Attitude controlled final braking with RCS (up to <2 m)</p>
- Landing on crushable structure at Meridiani Planum









Status

- > DHMR Test performed in the TAS-I Turin Autoclavet ~125 °C in dry nitrogen environment
- Sine Test Performed in ESTEC Test Facility
- Biosealing/Heatsealing verification performed before and after the above tests







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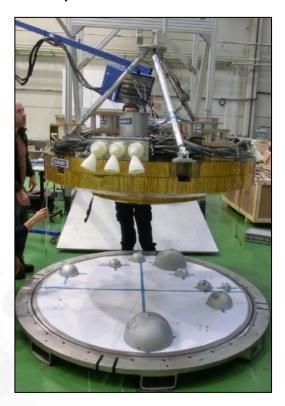
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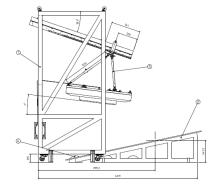




- Landing Impact tests
 - Crushing Test on Samples (performed)
 - Full Scale BB crush test campaign (performed)
 - Qualification Test on ESP QM
 - Test performed at subcontractor level (SENER)







Test on full scale BB

Credit of SENER



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2018 Mission

2018 Mission Objectives

Technological Objectives

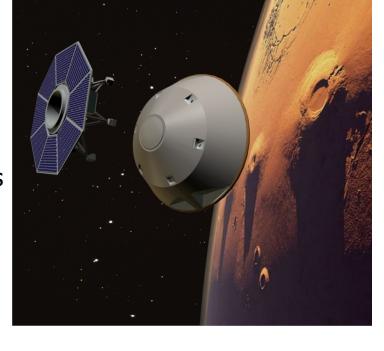
- Access to the sub-surface to acquire samples
- Sample acquisition, preparation, distribution, and analysis
 - Rover, Drill, ALD

Scientific Objectives

- Search for signs of past and present life on Mars
- Investigate the water/geochemical environment as a function of depth in the shallow subsurface (2018); Investigate Martian atmospheric trace gases and their sources (2016 and 2018)
 - **Rover, Drill, TGO**

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20 January 2012



2018 Mission Overview

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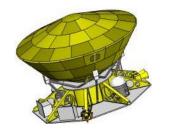
Launch: May 2018

Arrival: Jan 2019

Spacecraft Composite (Carrier Module + Descent Module + Rover Module)



Proton M/Breeze M





ROCC (incl. SOC-Altec)



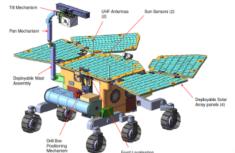






Lander Op. Center (ROS)







NASA DSN (in contingency cases)

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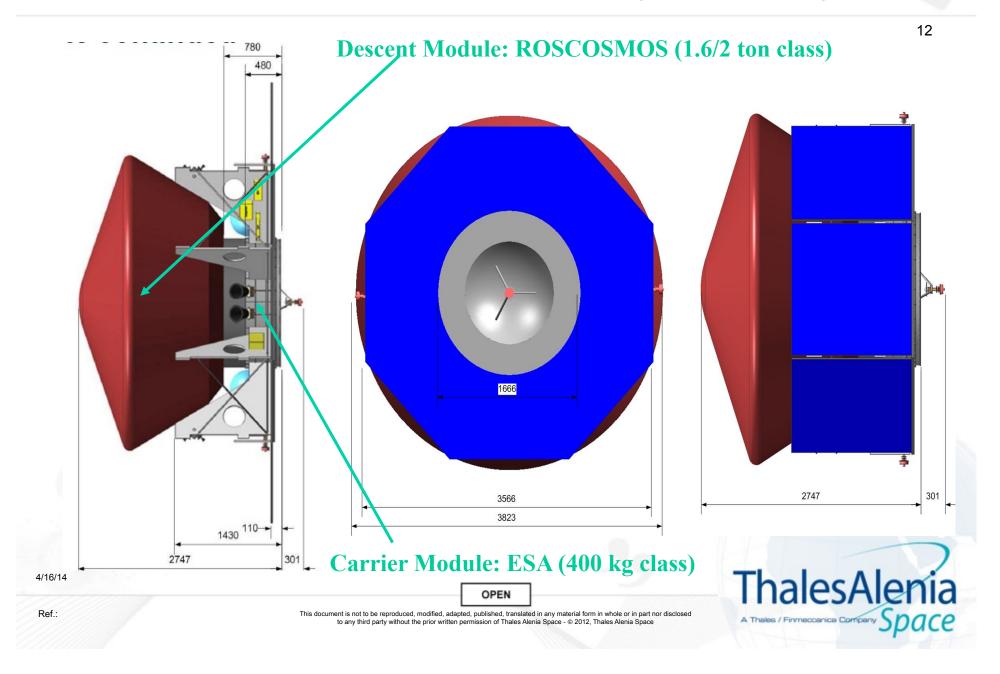
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20 January 2012

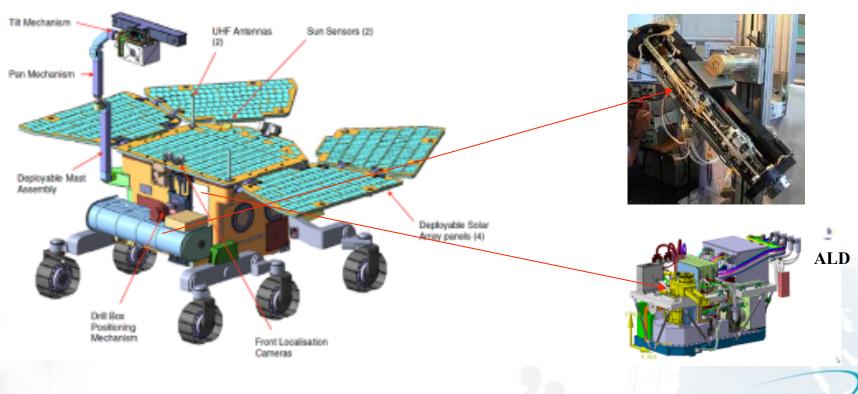
2018 mission - Spacecraft Composite



Rover: ESA (350 kg class)

Two Russian instruments (Infrared Spectrometer on Deployable Mast and Radiation Monitor on the RV body)

Drill



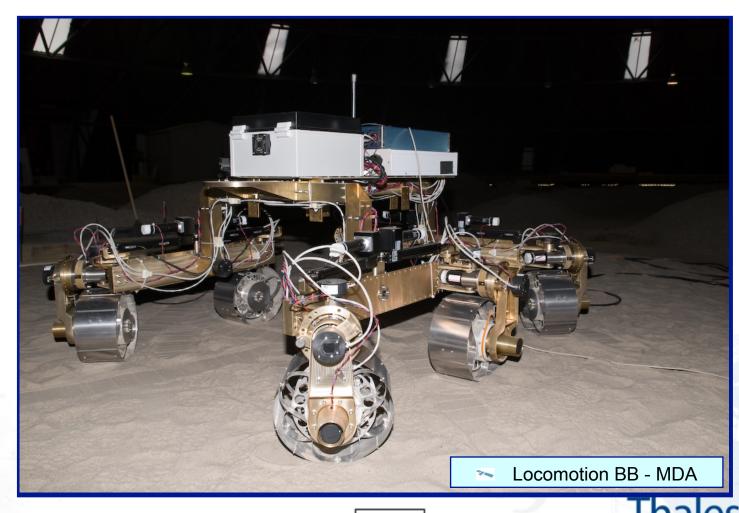
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24 Apr 2012

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Drill BB – Galileo Avionica

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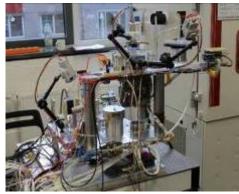


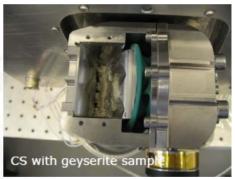


2018 Mission – SPDS Ongoing verification activities

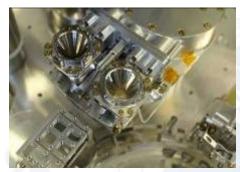
- The complete SPDS mechanism assembly has been assembled in order to perform functional and performance test of the mechanism chain in KT (D).
- E2E test campaign completed in both Laboratory and Mars-like conditions (temperature and pressure) with success.











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Turin 24



Milling Station BB – courtesy of Kaiser Threde



Dosing Station BB – courtesy of Kaiser Threde

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